Area of Operation II - Task A

Aircraft Flight Instruments & Navigation Equipment



Key References:

- Instrument Flying Handbook
- AIM

Content

- 1. Introduction
- 2. Instruments
 - Pitot-Static
 - Gyroscopic
 - Magnetic Compass
 - RMI & HSI
 - PFD
 - EIS Engine Information System
- 3. Navigation
 - ADF
 - VOR
 - DME
 - ILS
 - GPS
 - Transponder & ADS-B
- 4. De-lcing and Anti-lcing
- 5. Inflight Weather Services & Equipment

1. Introduction

- What: Learn about the aircraft's equipment used on Instrument Flying and navigation systems
- Why: Instrument Flying relies on aircraft's equipment and navigation systems, and knowing how they work, as well as their limitations, is critical for safe flying.

Instruments

o Airspeed, Attitude, Altimeter, Turn Coordinator (+ Turn Rate), HSI/HDG, Vertical Speed Indicator (VSI)

Navigation Types

<u>Ground based</u>:
 VOR, DME, NDB

 <u>Satellite based</u>: GNSS (GPS)



2. Instruments | Pitot-Static

- **Airspeed** Indicator, **Altimeter** and **VSI** (Vertical Speed Indicator)
- How it Works
 - <u>Altitude</u> calculated by expansion/contraction of an aneroid (<u>Static Port</u>) // <u>Not Corrected</u> for non-standard temperature
 - <u>VSI</u> calculated by the flow of a "calibrated leak" between the <u>VSI casing</u> and the <u>Static source</u>
 - <u>Airspeed</u> calculated by pressure differential (<u>diaphragm</u>) between the <u>Pitot</u> and <u>Static port</u> // Not true airspeed
 - G1000 does not use diaphragm/aneroid wafers, all is handled by the ADC (solid state components)

Potential Problems

- Subject to position errors (air flow)
- Pitot blockage: act as altimeter (increases w/ altitude), or goes slowly to zero if drain hole is open
 Action → Pitot heat on
- Static port blockage: altitude remains unchanged, Airspeed shows increasing as the airplane descends (pilot will tend to pull back, reduce power [§]). VSI zero. Action → Alternate static (altitude will read higher)



2. Instruments | Pitot-Static



- Does not correct for non-standard temperature
- Requires reference pressure on the Kollsman Window
- Alternate Static reads higher altitude

2. Instruments | Pitot-Static

• Effects on Altimeter

- When air is <u>colder than standard</u> \rightarrow the aircraft is at <u>lower altitude than altimeter indicates</u>
 - Rule of thumb: Altitude = 4% lower for every 10°C below ISA at low altitudes
- When flying from higher to lower pressure area without adjusting the altimeter:
 - The aircraft will be at lower altitude than altimeter indicates
- o High to Low or Hot to Cold → Look out Below
 - The higher you are, the higher the error will be (more air below you)





The altimeter is calibrated assuming standard atmosphere

2. Instruments | Gyroscopic

- Attitude Indicator, Heading Indicator, Turn Coordinator
- How it Works: Spinning discs mounted in a way to take advantage of the Gyroscopic properties
 - Power Sources: <u>Electrical</u> (turn coordinator) or <u>Vacuum</u> (attitude and heading indicators) Ο
 - Rigidity in Space: Gyro remains in a fixed position in space (attitude and heading indicators) Ο
 - <u>Precession</u>: Any applied force is felt 90° from that point in the direction of rotation (turn coordinator) Ο



Heading Indicator



Turn Coordinator (or Turn-and-Slip Indicator, less common)



Errors

- Slight nose up during rapid acceleration (and vice versa)
- Possibility of a small bank and pitch error after a 180° turn
- Tumble when bank >100° or pitch >60°

Errors

- · Precession results in heading drift
- (+Earth rotates at 15° per hour)
- Check and reset the heading indicator every 15 minutes

How does it work

- Yawing produces a force in the horizontal plane tilting the gyro left or right. <u>Spring returns to neutral.</u>
 Turn-and-Slip: Gyro is vertical (senses YAW), shows Rate/Direction of turn (<u>does not</u> show bank/roll)
 Turn Coordinator: Gyro is tilted 30° (senses YAW and ROLL). Shows initially "Rate of Roll" and then <u>RoT</u>

2. Instruments | Gyroscopic

Vacuum System (C172S G1000)

- 1 Air Filter, 1 Vacuum regulator
- 1 Engine-driven vacuum pump
- <u>Standby</u> Attitude Indicator
- Vacuum transducer → "Green" pump pressure 4.5-5.5 inHg
 - LOW VACUUM annunciator goes off if below 3.5inHg



C172S Vacuum System (Backup Attitude Indicator)





2. Instruments | Magnetic Compass

Compass Errors

- Variation: difference between magnetic north (compass) and true north
- o <u>Deviation</u>: interference w/ aircraft's equipment. Corrected by the compass card.
- <u>Magnetic</u> dip: earth's magnetic flux is 3D, so the compass will try to dip to align
- Turning: to the N (compass lags), to the S (it accelerates). UNOS → 15° + Half of the Latitude
- o <u>Acceleration</u>: when flying on **E/W heading**, compass needle points towards the North when accelerating. ANDS
- o Oscillation: float assembly bouncing around
- NOTE: limitations steep bank (>18⁰) shows erratic readings





2. Instruments | RMI & HSI





• RMI (Radio Magnetic Indicator)

- Bearing indicator (ADF, VOR, both) overlayed on a heading indicator
- Heading (Compass Card) is automatically rotated by a Flux Valve (earth's magnetic field)

HSI (Horizontal Situational Indicator)

- Combines a heading indicator with a Navigation source (e.g. VOR, ILS, etc)
- Compass card automatically aligns to magnetic north (flux valve)
- Shows deviation from selected radial/bearing

2. Instruments | PFD

G1000

- GDU (Graphic Display Unit): <u>PFD</u> and <u>MFD</u>
- GIA (Integrated Avionic Unit): 2 units, one associated to each GDU
 - <u>GPS receiver</u>, <u>COMM/NAV receivers</u> and connect all systems
- ADC (Air Data Computer): <u>Airspeed, Altitude, VSI, OAT</u>. Behind the MFD
- AHRS: <u>Attitude</u>, <u>Rate-of-Turn</u>, <u>Slip/Skid</u>. Solid-state (or MEMS) gyro/accel.
- Magnetometer (GMU):
 - Detects magnetic field info and send to the AHRS for <u>Heading</u>
- Autopilot Servos (GSA)
- Autopilot (GFC700)
- Audio Panel
- Transponder
- Eng/Airframe unit

Note: <u>G3X Touch</u> works with LRUs in multiple configs





2. Instruments | Engine Information System

- EIS ٠
 - Monitoring Instrument that provides information about the aircraft's engine, Ο fuel and electrical systems
 - System can include: Ο
 - LCD Displays Ο
 - Warning signals (messages, blinking lights, etc) Ο
 - Alarms Ο
 - Sensors Ο

C172S Ο

- **EGT**: Exhaust Gas Temperature Ο (1250F-1650F)
- **CHT**: Cylinder Head Temperature Ο (200F-500F)





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3. Navigation | ADF (Automatic Direction Finder) & NDB

• What

- Points to a station (being decommissioned)
- NDB operates between 190-535 KHz and transmit radio energy in all directions
- Volumes (radial dist.): HH (75nm), H (50nm), MH (25nm), Compass loc. (15nm)

Components

- NDB (Ground), ADF (Aircraft)
 - ✓ ADF = Automatic Direction Finder (2 antennas, Receiver and Instrument)
 - ✓ ADF operates between 530-1700 kHz and can be tuned to AM stations
- o <u>Limitations</u>: subject to interference
- Instrument
 - Fixed Card (top)
 - Movable Card (middle)
 - RMI (Radio Mag Indicator, bottom)





Automatically update hdg (Fluxgate)



AIM 1-1-2

Relative Bearing 135°





3. Navigation | VOR (VHF Omni-Directional Range) AIM 1-1-3

What:

• Allow the pilot to fly <u>magnetic courses</u> or <u>identify radials</u> by using stations as reference (radio signal emitted from the ground)

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• 3 Classes: Terminal, Low, High → Based on Service Volumes

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• Different variations: VOR, VOR/DME (VOR+DME), VORTAC (VOR + TACAN)







Service Volumes expanded under MON

3. Navigation | VOR (VHF Omni-Directional Range) AIM 1-1-3

How it Works

- <u>Station</u>: emits a rotating signal (30 times/sec) + ref signal when crossing 360 radial
- <u>Aircraft Receiver</u>: calculates difference between both signals to find its radial
- <u>108.0 to 117.95 MHz</u> (excl LOC frequencies of 108.10-111.95 w/ odd tents)
- Limitations: Line-of-Sight, Cone of confusion, Reverse Sensing
- Minimum Operational Network (MON): VOR within <u>100nm</u> (>5000ft)
 - ✓ Reduction from 896 to 590 VORs, expanding service volumes



VOR Receiver Checks (§91.171)

Every 30 days for IFR Navigation

- ∨OT (360 From) → ±4°
- Dual VOR → ±4°
- Ground Checkpoint → $\pm 4^{\circ}$
- Repair Station $\rightarrow \pm 4^{\circ}$
- VOR Airborne Checkpoint → $\pm 6^{\circ}$
- Above landmark within 20nm of a VOR, low altitude $\rightarrow \pm 6^{\circ}$ of selected radial
- Record:
 - ✓ Date
 - ✓ Error
 - ✓ Place
 - ✓ Signature

3. Navigation | VOR (VHF Omni-Directional Range)



3. Navigation | VOR (VHF Omni-Directional Range)

Steps for VOR Navigation

- 1. Tune to the VOR frequency and identify it
- 2. <u>Turn the OBS</u> to determine the <u>bearing</u> you are in (needle centered w/ TO indication)
- 3. <u>Turn the aircraft until your heading aligns to the VOR course selected</u>
- 4. If you want to cross the VOR in a different radial, turn the OBS to the desired bearing
- 5. Turn the airplane to intercept the radial that leads to the desired bearing
- 6. Change station at the changeover point
- 7. Wind correction:
 - If drifting, turn 10° into the wind
 - Adjust WCA depending on the results (under/over-shoot)

Avoid reverse sensing

- If flying to the VOR, flag w/ TO
- If flying from the VOR, flag w/ FROM
- Note: no reverse sensing in the G1000

Flying heading 030 towards the 240 radial (intercept) and fly bearing 060

OBS



3. Navigation | DME (Distance Measurement Equipment) AIM 1-1-7

What

- Provides distance from a station
- 2 classes: <u>Low</u> and <u>High</u> depending on Service Volume

How it Works

- Airplane interrogates, station responds → Receiver calculates distance
- <u>UHF</u>: Operates in the 962-1213 MHz frequencies (paired w/ VOR freq)
- <u>Limitations</u>:
 - o Line-of-sight
 - o Slant Range
 - \checkmark The higher you are and the closer you are \rightarrow the higher the error
 - ✓ Negligible 1nm horizontal per 1000ft alt
- DME receiver can be replaced by a RNAV system (G172 doesn't have a DME)
- DME receiver (or RNAV) is required above FY240 (FAR 91.205)





3. Navigation | ILS (Instrument Landing System) AIM 1-1-9

What: Electronic system providing horizontal & vertical guidance for landing

How it Works

- Localizer (LOC, VHF), Glide Slope (GS, UHF), Marker Beacons and ALS
 - LOC frequencies: 108.1 111.95 MHz with odd tenths only
 - Glideslope: Generally 3° glide path, 1.4° thick, tied to the LOC frequency
- Minimums (DH/RVR): CAT-I (200ft/1800-2400), CAT-II (100ft/1200), CAT-III (up to 0) *CAT II & III requires specific certification and equipment
- <u>Limitations</u>: subject to obstruction, interference (surface vehicles/aircraft can disturb the signal), and false echoes for GS



Marker Beacon

- Outer (OM): 4-7 NM out/GS intercept
 Blue/Purple, low pitch, dashes
- Middle (MM): 3,500' out/200'
 Amber, intermediate pitch, dots/dashes
- Inner (IM): DH on a Cat II approach White, high pitch, dots
- Compass Locator: With OM/MM

3. Navigation | ILS (Instrument Landing System)

Approach Light Systems

- Transition to Visual Flight [AIM 2-1-1]
- ALSF: Approach Light System with Sequenced Flashing lights
- **SSALR**: Simplified Short Approach Light System with Runway alignment indicator lights
- **MALSR**: Medium intensity Approach Light System with Runway alignment indicator lights
- **REIL**: Runway End Identification Lights
- **MALSF**: Medium intensity Approach Light System with Sequenced Flashing lights (& runway alignment)
- ODALS: Omnidirectional Approach Light System



3. Navigation | ILS (Instrument Landing System)

Localizer Receiver

- Same as a VOR receiver
- <u>Navigation</u>:
 - Like a VOR, but more sensitive (2.5° full deflect. vs 10°)
 - Rotating the OBS has no effect
 - Back course has opposite sensing
- <u>Flying</u>
 - \circ $\,$ Center the LOC, fly the inbound course
 - o Apply drift corrections



Glideslope Receiver

- Auto tuned with the localizer frequency
- <u>Navigation</u>:
 - Needle/diamond represent the glideslope
 - On Glidepath: Needle or diamond centered
- <u>Flying</u>
 - o Set pitch & power as GS is intercepted
 - Small pitch adjustments to maintain slope



GNSS (Global Navigation Satellite System) – AIM 1-1-17

- <u>Types of systems</u>: GPS (US), Galileo (EU), GLONASS (Russia), BeiDou (China)
- <u>GPS Space Segment</u>: constellation of <u>min 24 satellites</u> (current 31) orbiting earth at <u>11,000 miles</u>
- <u>GPS Control Segment</u>: position/clock accuracy 1 Master control station, 5 monitoring stations, 3 ground antennas
- Number of satellites required for triangulation: 3 (2D), 4 (3D), <mark>5 (RAIM)</mark>, 6 (fault exclusion)





SBAS (Satellite Based Augmentation System) – AIM 1-1-18

- <u>Types of systems</u>: WAAS (US), EGNOS (UE), SDCM (Russia, under dev.), etc
- <u>WAAS</u> (Wide Area Augmentation System): improves accuracy of GPS signal
 - ✓ <u>38 Stations</u> (WRS): compare GPS signal w/ their own surveyed location
 - ✓ <u>3 Master Stations</u> (WMS): receive data from the WRSs, calculate the GPS errors and transmit to 6 ground uplink stations
 - ✓ <u>3 Geostationary Satellites</u> receive the signal from the uplink stations and broadcast to WAAS receivers, which correct the GPS signal





Using your GPS (multiple models available) in an MFD

- <u>Steps</u>: (1) Load Waypoints → (2) Follow the Magenta CDI (like a VOR) → (3) Check the moving maps (MFD)
- Be careful with the "Direct To" GPS will not avoid terrain for you



RAIM (Receiver Autonomous Integrity Monitoring) – AIM 1-1-17(a3)

• Required to ensure satellite signals meet integrity req → Need at least 5 satellites (6, or 5 + baro aid, for fault exclusion)

GPS Equipment – AIM 1-1-18(c), AIM 1-1-17(b2a2), AIM 1-1-3(f3b2)

- <u>TSO 129/196 (non-WAAS):</u>
 - Can fly RNAV(GPS) approaches to <u>LNAV minima</u> (LNAV/VNAV with baro-aided systems)
 - When filing an <u>alternate</u> airport, at least one of them (destination or alternate) must have a <u>non-GPS approach</u>
 - <u>Needs to confirm RAIM availability</u> as part of pre-flight.
 - Under IFR, must be equipped with an alternate means of navigation (if lose RAIM, must revert to VOR navigation)
- <u>TSO 145/146 (WAAS):</u>
 - Can fly the approaches above, plus <u>LPV</u>, LP, and LNAV/VNAV minima w/o baro-aided system.
 - When filing an alternate airport, there are no restrictions on type of approach, both can be GPS
 - <u>RAIM is not required</u> (unless if lose WAAS signal).
 - WAAS users flying under Part 91 are not required to carry VOR avionics for IFR/GPS operations

GPS may be used as a substitute when:

- Determining position over a DME fix and/or Flying a DME arc
- Navigating To/From an NDB or Compass Locator, or Holding over an NDB/Compass Locator
- Determining position over a fix defined by an NDB or Compass Locator bearing crossing a VOR/LOC course
- Can use GPS to fly a VOR approach as long as the VOR is operational and monitored [AIM 1-2-3 (c5)]

3. Navigation | Transponder w/ Altitude Reporting

Allows ATC to see where the aircraft is

- Unique squawk code assigned by ATC to the aircraft can be monitored by them throughout the controlled airspace
- o <u>Other codes</u>: 7500 (hijack), 7600 (lost comms), 7700 (emergency), 1200 (VFR)

• Modes and Functions

- Mode A: only transmits a four-digit squawk code and position ("Mode 3" is the equivalent for Military use)
- Mode C: In addition to Mode A, it also transmits the aircraft's pressure altitude
- Mode S (Select): In addition to Mode C, it also transmits speed, heading and ICAO address. TCAS and ADS-B.
- o <u>Ident</u>: Ident is a function of a transponder that, when activated, allows ATC to identify your aircraft on its radar screen.
- <u>Status</u>: OFF, SBY (standby, not transmitting), ON (operating on Mode A/3), ALT (operating on Mode C)

Altitude Validation

• ACT must verify (by regulation) \rightarrow inquire pilot's altitude, <u>it must be within 300ft</u> of what the controller sees

Maintaining

- Inspection required every 24 calendar months) [91.413]
- If the aircraft has an operational transponder, it must be on and transmitting per 91.215(c)



3. Navigation | ADS-B

Automatic Dependent Surveillance Broadcast

- ADS-B Out: Broadcasts GPS location, altitude, groundspeed, and more to ground stations & other aircraft
- ADS-B In: Pilots can see what controllers see, as well as weather (if installed)
 - ✓ **TIS-B** (<u>Traffic</u> Information, 978MHz (UAT)/1090ES): ADS-B Out equipped aircraft + radar (~30nm, ±3,500ft)
 - ✓ FIS-B (Flight Information, 978MHz (UAT) only): weather products, clouds, icing, metar, airmet/sigmet, etc
- Requirements for ADS-B Out (91.225)
 - Equipment satisfies 91.227 and 91.225(a)(1) and (b)(1) note more options below 18,000ft MSL
 - Class A, B, and C airspace (and <u>above Class B and C</u>, up to 10,000ft MSL) and <u>Inside Mode-C</u> veil up to 10,000ft MSL
 - Class E airspace at and above 10,000ft MSL, excluding below 2,500ft AGL
 - ✓ At and above 3,000ft MSL over the Gulf of Mexico within 12nm of the coast

Use portable ADS-B In receivers if aircraft not equipped





3. Navigation | Automatic Pilot & FMS

FMS (Flight Management System)

- Receives inputs from a variety of sensors and has a pre-loaded database of NAVAIDS, Airways, etc
- Employs calculations to determine the action/guidance on all phases of flights (e.g. directs the auto-pilot)

• Autopilot

- o Single Axis (hold wings level), Two Axis (adds pitch control), Three Axis (adds yaw control)
- \circ Modes
 - **ROL/PIT** → Maintain current bank/pitch attitude
 - HDG → Maintain specified heading
 - NAV → Follows the Navigation source
 - APR → Navigation + GS on approaches
 - ALT \rightarrow Hold altitude
 - VS → Recommended for Descent. Defines Vertical Speed (RPM)
 - IAS (or FLC) → Recommended for Climb. Defines Airspeed (Kts)
 - ∨NAV → follows descent paths (navigation)
- Examples: GFC500, GFC700





4. De-Ice and Anti-Ice

- Anti-ice prevents ice, De-ice removes ice
- Airfoil
 - 1 Deice Boots → Inflatable rubber boots bonded to the leading edge of the wing
 - <u>Thermal Anti-Ice</u> \rightarrow Hot air from the engine compressor
 - <u>Weeping Wing</u> → Antifreeze weeps through small holes in the leading edge
- Windscreen and Propeller
 - <u>Alcohol</u> Used to prevent ice build up on windscreen / propeller
 - <u>Electric Heat</u> Wires imbedded in the windscreen / propeller anti-ice
- **TKS System** (Tecalemit-Kilfrost-Sheepbridge Stokes)

"Weeping wing", but a system that covers way more than the wing

• C172S

(2)

- Flying into known icing condition is prohibited in the C172 by the POH
- o <u>C172 does not have deicing</u>, only <u>minimal anti-icing</u>
- **Pitot heat** \rightarrow electrical heating element. Must be on if OAT <40°F (4.4°C)
- Windshield defrost → heated air routed to the top of the panel



5. Inflight Weather Services & Equipment

- Weather Datalink
 - FIS-B (Broadcasted by <u>Ground Stations</u>, 978MHz (UAT)): clouds, icing, metar, radar, airmet/sigmet, etc
 - Sirius XM: via <u>satellite</u>, requires subscription
- Onboard Doppler Radar & Stormscope
 - Radar works with UHF radios echoes
 - Lightning: Detects discharges within 200nm (WX-500 stormscope)







Questions?

